



# The Glacier and Land Ice Surface Topography Interferometer (GLISTIN): A Novel Ka-band Digitally-Beamformed Interferometer

June 29, 2006

Delwyn Moller, Brandon Heavey, Richard Hodges, Sembiam Rengarajan, Eric Rignot, Francois Rogez, Gregory Sadowy, Marc Simard, Mark Zawadzki

Jet Propulsion Laboratory
California Institute of Technology





### A Ka-band Digitally Beamformed Radar Interferometer for Topographic Mapping of Glaciers and Ice Sheets



### Objective

PI: Delwyn Moller / JPL

A high accuracy radar for ice topography mapping (both icesheets and glaciers) over a wide swath with sub-seasonal repeat intervals. The instrument is a Ka-Band digitally-beamformed interferometric synthetic aperture radar. The use of millimeter-wave signals increases accuracy, decreases mass and reduces snow penetration. Elevation digital beamforming preserves swath yet maintains high antenna gain on receive



Conceptual rendition of the deployed cross-track interferometer

#### Approach

- A mission design and trade study will be performed to define the antenna requirements
- Integration of the antenna from radiating elements to digitization
  - development of lightweight radiating elements
  - development of a small digital receiver
  - development of a phase-stable antenna array
- Demonstration of entire array will verify antenna, calibration and beamforming concept.

Co-Is: Greg Sadowy, Eric Rignot, Mark Zawadzki (JPL)

### Key Milestones

· System/Science Requirements Document	4/06
<ul> <li>Radiating Element Design and Test</li> </ul>	1/07
· L-Band Receiver Fabrication Complete	7/07
· Demonstrate Ka-band Downconverter	8/07
· Science Impact Assessment Report	1/08
· Radar Experimental Demonstration	8/08

http://esto.nasa.gov



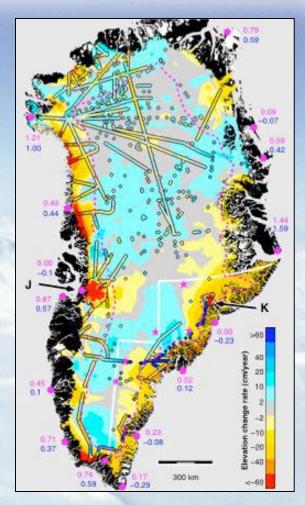


# Scientific Motivation (1)



Fundamental question: "What changes are occurring in the mass of the Earth's ice cover"

- The Greenland and Antarctic ice-sheets together hold enough ice to raise global sea level by 80m
- The annual exchange of mass on the ice sheets is equivalent to 8mm/yr sea level=>any fluctuations in that level of exchange is significant on the global scale
- Changes in the polar region are rapid (years not centuries) and significant (meter-scale not millimeters)



Greenland ice changes 1997-2003. (Krabill, *et al*)





## Scientific Motivation (2)



Current technologies limited to profiling airborne and spaceborne sensors (radar and lidar altimeters)

- Profiling techniques limited in swath coverage
- Airborne sensors impractical for global coverage
- IceSAT is an effective sensor, but unable to penetrate clouds and limited in swath

A radar mapping sensor impervious to cloud cover and is able to provide swath measurements at a variable spatial resolution consistent with the differing requirements of glacial vs ice-sheet mapping.

Topic	Req #	Requirement
Coverage	L2-CR1	Monthly to bi-monthly coverage of glaciers and ice sheets
	L2-CR2	Complete coverage of Greenland and Antarctica with North Pole hole.
Glaciers	L2-GR1	100 m x 100 m horizontal resolution for glacier height measurements.
	L2-GR2	100 m x 100 m horizontal posting for glacier height measurements.
	L2-GR3	1 m relative height error for glacier measurements
Ice Sheet	L2-IR1	1.0 km x 1.0 km horizontal resolution for ice sheet measurements.
	L2-IR2	1.0 km x 1.0 km horizontal posting for ice sheet measurements.
	L2-IR3	10 cm relative height error for ice sheet measurements.





### **Orbit Selection**

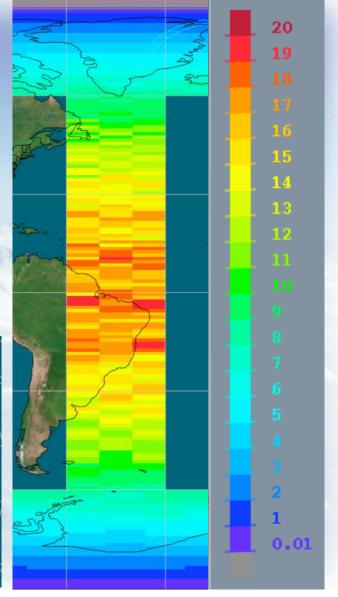


- Given:
  - Altitude close to 600 km
  - Incidence angle range: 18.6 to 25.2 deg.
  - => Swath extends from 184 to 256 km on one side of the ground track
- Requirements:
  - Coverage of South Pole
  - Accessibility of all land-ice.
  - Hole around North Pole is acceptable.
  - Requirement: Monthly to bi-monthly acquisitions
- Best altitude/orbit count: 605.7 km, 593 orbits per repeat cycle, 40 days to repeat
- Chosen orbit: 92 deg inclination, left looking (based on above & crossover separation)

Near right: Sun-synchronous orbit is not acceptable: gap from -85.1 to South Pole

Far right: Average temporal gap between successive revisits









# System Design Overview





Ka-band Receiving Elements	Ka-band Digital Receiver
Ka-band Receiving Elements	Ka-band Digital Receiver
Ka-band Receiving Elements	Ka-band Digital Receiver
Ka-band Receiving Elements	Ka-band Digital Receiver
Ka-band Receiving Elements	Ka-band Digital Receiver
Ka-band Receiving Elements	Ka-band Digital Digital Data Receiver Multiplexer
Ka-band Receiving Elements	Ka-band Digital Receiver
Ka-band Receiving Elements	Ka-band Digital Receiver
Ka-band Receiving Elements	Ka-band Digital Receiver
Ka-band Receiving Elements	Ka-band Digital Receiver

- A Ka-band InSAR system can meet the science requirements
  - 70 km swath
  - 600 km altitude
  - 10 cm height error at 1 km x 1 km resolution
- Ka-Band is chosen for two reasons:
  - 1. Insensitive to cloud cover, yet minimizes snow penetration
  - 2. A high (mm-wave) frequency allows us to maintain high accuracy with a smaller baseline
- Problems with previous mm-wave InSAR
  - high transmit power requirements and
  - limited swathwidth (large antennas required for high-gain leading to reduced swathwidth).
- We apply digital-beamforming to a large array on receive to maintain high gain, yet at achievable transmit powers.
  - Approximate 12-fold saving in transmit power for same swath-width.





# **Key System Parameters**





Ke hand Davidson Floresate	Ka-band Digital
Ka-band Receiving Elements	Receiver
Ka-band Receiving Elements	Ka-band Digital
	Receiver
Ka-band Receiving Elements	Ka-band Digital
	Receiver
Ka-band Receiving Elements	Ka-band Digital
rka-band rkecelving Elements	Receiver
Ka-band Receiving Elements	Ka-band Digital
Na-band Necelving Elements	Receiver
Ka-band Receiving Elements	Ka-band Digital Data Digital Data
Na-band Necelving Elements	Receiver Multiplexer
Ka-band Receiving Elements	Ka-band Digital
Na-band Necelving Elements	Receiver
Ke hand Dessiving Flaments	Ka-band Digital
Ka-band Receiving Elements	Receiver
Ka haad Daad daa Elamada	Ka-band Digital
Ka-band Receiving Elements	Receiver
Ka-band Receiving Elements	Ka-band Digital
Na-band Necelving Elements	Receiver

Parameter	Units	Quantity
Peak Transmit Power	kW	1.5
Frequency	GHz	35.75
Bandwidth	MHz	40
Antenna Length	m	4
"stick height"	m	0.063
Number of sticks	#	16
Total array height	m	1.01
Pulsewidth	us	25
Prf	kHz	4
Interferometric baseline	m	8
Polarization	-	Horizontal
Swath-width	km	70
Incidence angle range	deg	18.6 - 25.2
SNR over swath	dB	3.0-10.0





# System Design Issues

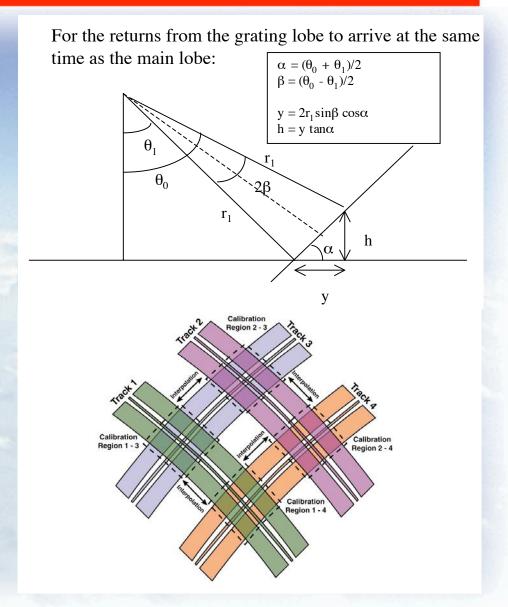


### Grating Lobes:

- have limited the number of "sticks" to 16 creating an inter-stick spacing of  $\sim 7.4\lambda$
- we have grating lobes at ~+/- 7 degrees.
- When beamforming off-nadir the grating-lobe levels become significant
- we are able to range-gate out the ambiguous returns.

# • Stringent Antenna Location Knowledge Requirements

- requirements are extremely stringent: i.e. baseline roll must be known to 0.25 arcsec
- WSOA demonstrated that such stringent requirements could be met through cross-over calibration techniques
- We can apply this technique to GLISTIN (a topic of ongoing work)



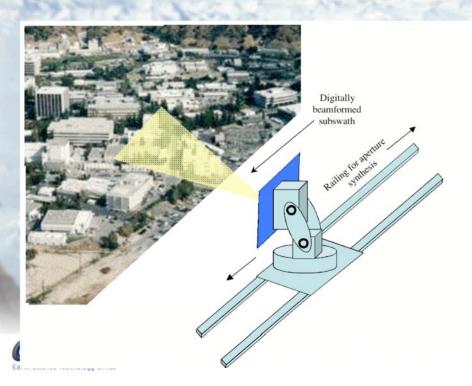




### Demo Overview



- The main objective of this program is to demonstrate key technology (Ka-band DBF antenna) & associated processing
- Where possible the requirements for the key technology items are that of the spaceborne design.



### **Demo Overview**

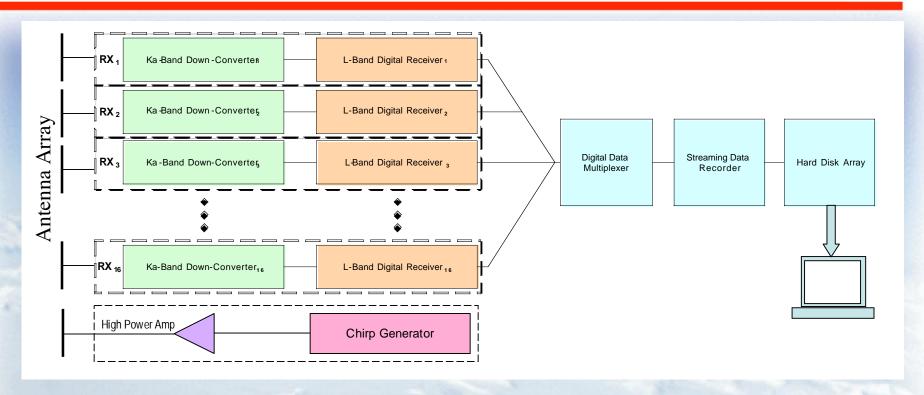
We will use the JPL mesa antenna range to characterize and then demonstrate the calibration and performance of the antenna array and the entire end-to-end concept for the DBF interferometric image synthesis.

- The simple radar system will be mounted on a railing overlooking JPL and rotated in azimuth on the positioner to form an image
- DBF will be used to scan in elevation on a fine-scale coupled with elevation scanning on a coarse-scale using the positioner
- A 0.5m interferometric baseline will be achieved by halving the effective aperture - sufficient at the close range of operation



# Demo Block Diagram & Technology Matrix





Subsystem / Component	Key Technology
Antenna Aperture	Slotted Waveguide
Ka-band Electronics	mm-wave hybrid microelectronics
L-band Digital Receiver	Atmel ADC + Xilinx FPGA
Data Acquisition	COTS



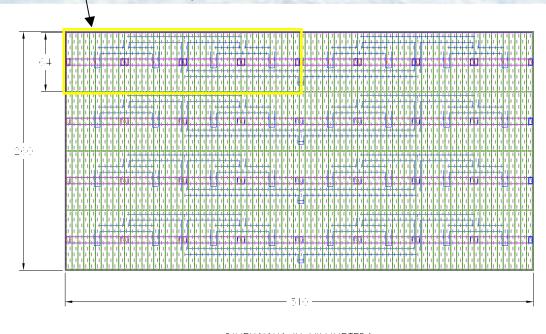


# Technology Detail: Antenna



# Demonstrate a 1 m x 1m Array (ie 1/4 of the spaceborne antenna size) $\rightarrow$ a $160 \times 160$ slot array

- pushes fabrication limits for such a large array at Ka-band
- vendor recommended subdividing fabrication modules to 40x80,
   requiring a total of eight for assembly of the 1m x 1m array
- make two 10x40's this year to work out fab and electrical details





DIFFERENCE OF THE FILLINGTERS

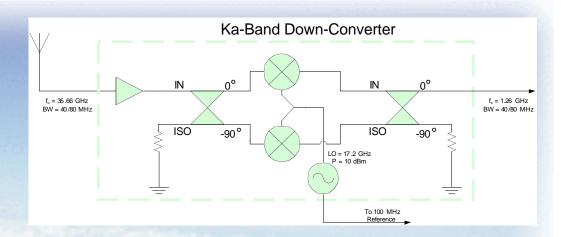


# Technology Detail: Digital Receiver



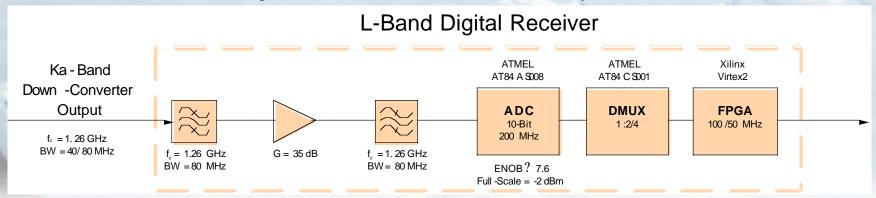
#### Ka-Band Down-Converter Element

- Down-Converts Input Ka-Band
   Signal to L-Band
- Utilizes Image-Rejection, Even-Harmonic Mixer
- To be Procured as a Single,
   Connectorized Component



### • L-Band Digital Receive Element

- Filters Out Undesired Portion of Input Spectrum
- Adjusts Signal to Proper Level for Sampling
- Bandpass Sampling of L-Band (Direct Digitization)
- DMUX Data to Slow Operating Frequency Requirement of FPGA
- Utilize FPGA for Buffering of Data and to Communicate with Data System







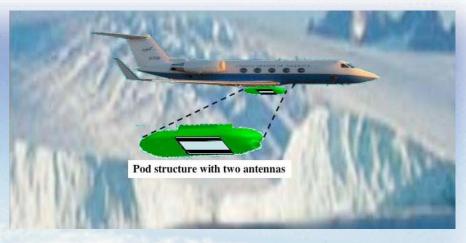
### Further Opportunities



### International Polar Year Proposal

**Investigators:** Delwyn Moller, Greg Sadowy, Eric Rignot, Marc Simard, Scott Hensley (JPL)

**Collaborators:** Prof. Koni Steffen (Univ. of Colorado), Dr. William Krabill (NASA/GSFC)



A proposal to the International Polar Year NRA has been submitted for which we propose to adapt the ESTO UAVSAR system to operate in a single-pass interferometric mode. Our approach is to heavily leverage ESTO technology already developed:

- An ESTO IIP-developed Ka-band upconvert/downconvert chain is added to the L-band system
- The two polarimetric channels of UAVSAR are used for the two interferometric channels
- The L-band antenna panel is replaced with two 1m slotted waveguide antennas using the same technology as the GLISTIN antennas. The baseline is limited by the pod accommodation to approximately 0.5m
- The UAVSAR data-acquisition system and processing infrastructure is already capable for this application

#### **Objectives:**

- 1. To demonstrate single-pass Ka-band interferometry and calibration
- 2. To characterize the penetration depth of Ka-band into snow cover as a function of snow wetness and incidence angle. Critical to the feasibility of a future GLISTIN ice-topography mission
- 3. To collect a cohesive mosaicked topographic map over Jakobshavn glacier (Greenland) in collaboration with IPY scientists





# Backup Material





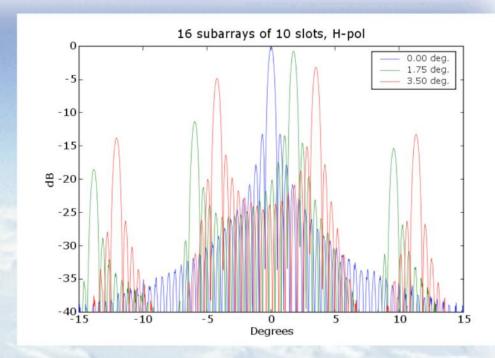




# Grating Lobes due to "Inter-stick" spacing



- In order to avoid grating lobes an array must be critically sampled in space (i.e. interelement spacings of λ /2 or less)
- An array response is the product of
  the element pattern times the array
  pattern therefore the element pattern
  can be used to reject grating lobes in
  non-critically sampled arrays.
   However if the array is steered offnadir, the element pattern stays static
  but the array pattern shifts decreasing
  the grating lobe rejection
- In the case of GLISTIN we have limited the number of "sticks" to 16 creating an inter-stick spacing of ~7.4 λ => we have grating lobes at ~+/- 7 degrees. When we digitally beamform off-nadir the grating-lobe levels become significant however we are able to range-gate out the ambiguous returns.





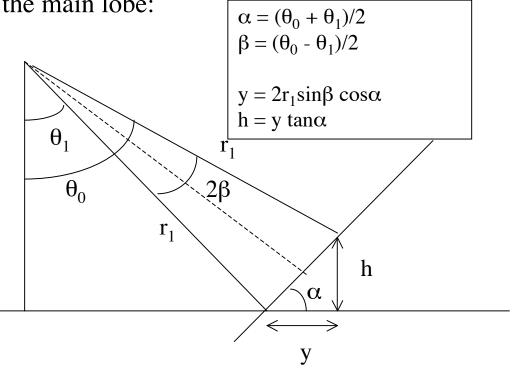


### **Range Ambiguities Due to Surface Slopes**



For the returns from the grating lobe to arrive at the same

time as the main lobe:



Number of Slots (0.74λ)	Grating lobe separation	Critical Slope (@ max $\theta_0$ )	Critical Distance	Δheight
10	7.7 °	18.2 °	42.4 km	13.9 km

Assumptions:  $\theta_0 = 23^\circ$ , h = 600 km





### Systematic Errors & On-orbit Calibration



# Systematic error budget required to meet science requirements is extremely challenging:

#	Requirement Statement	Value	Unit
L3-SY1	The baseline roll shall be known to within	0.025	arcsec
L3-SY2	The antenna phase mismatch shall be known to within	0.025	deg
L3-SY3	The receiver phase mismatch shall be know to within	0.04	deg
L3-SY4	the basline dilation shall be known to within	2.00	um
L3-SY5	the ADC system timing shall be accurate to	0.035	ns
L3-SY6	The spacecraft attitude shall be known to within	0.025	arcsec
L3-AP1	Each GLISTIN antenna shall have a boresight elevation pointing angle of	20.5 +/-0.5	deg

The Wide Swath Ocean Altimeter (WSOA) demonstrated through significant analysis and simulation that such stringent requirements could be met through the use of onorbit self calibration scheme using nadir altimeter tracks

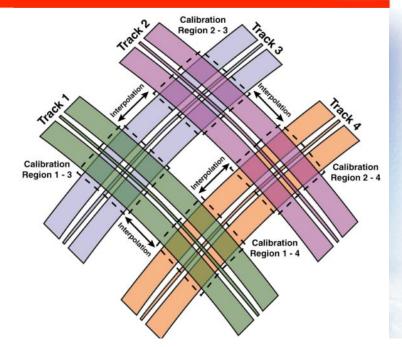


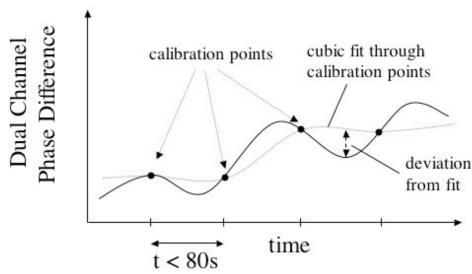


### WSOA Calibration Strategy



- WSOA assumed no change in the (incoherent) height between ascending & descending passes
  - Time between ascending / descending cross-over points less than +/- 5 days
- Use the ascending altimeter track (not subject to roll & phase errors) to calibrate the descending WSOA interferometer track at each crossover
- Interpolate between crossovers (no more than 80 seconds apart)
- Resultant requirement is on the residual after removal of the interpolated fit





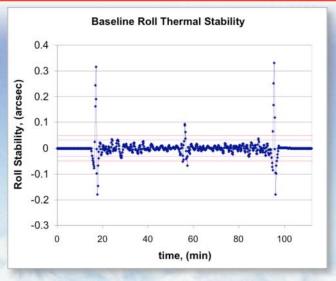




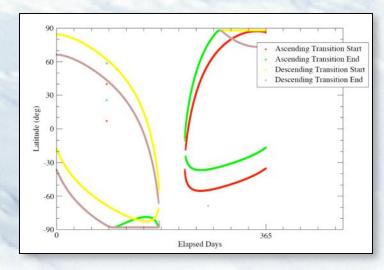
### WSOA Lessons Learned & Go Forward Plan



- During WSOA we learned that the stringent residual systematic error requirements were typically met (with 80 second crossovers) at all points outside of eclipses
- During the next phase of the GLISTIN IIP, work will focus on:
  - 1. Understanding "time between crossovers" based on re-analysis of WSOA deformation data, both near & far from eclipse points
  - 2. Determining appropriate calibration approach over ice using combination of sea surface at coast, data from previous passes over different points on the ice sheet, etc.
- Data near eclipse points can conceivably be neglected if self calibration is not feasible
  - This was the approach adopted by WSOA



Profile of baseline roll after fit removal showing jumps at eclipse points



Eclipse latitude over 1 year

